10/14

Aer-MA-610/1413

FIFTH ENDORSEMENT on VA-212 AAR ser 1-57 concerning FJ-4B BUNO 141483 accident occurring 22 October 1957, pilot (b) (6) 12 AUG 1958

From: Chief, Bureau of Aeronautics To: Chief of Naval Operations

Via: Commander, U.S. Naval Aviation Safety Center

Subj: Aircraft Accident Report; forwarding of

1. Forwarded.

b) (5)

(b) (6

by direction

Copy to:
COMNAVAIRPAC
CINCPACELT
COMFAIRALAMEDA
COMCARAIRGRU-21
CO VA-212

FF4-1/A25 Serial: 80/

522

SPECIAL HANDLING REQUIRED IN ACCORDANCE WITH PART VII OPNAVINST 3750.6B

1 6 JAN 1958

FOURTH ENDORSEMENT on VA-212 AAR ser 1-57 concerning FJ-4B BUNO 141483 accident occurring 22 October 1957, pilot (b) (6)

From: Commander Naval Air Force, Pacific Fleet

To: Chief of Naval Operations (OP-57)

Via: (1) Chief, Bureau of Aeronautics (MA-61)

(2) Director, U. S. Naval Aviation Safety Center

Subj: VA-212 AAR ser 1-57

Ref: (a) BUAER msg 072147Z of NOV 1957

l. Forwarded, concurring in the conclusion contained in the third endorsement, the recommendations of the Aircraft Accident Board and the remarks contained in the second endorsement. Previous endorsers are directed to change their records accordingly.

(b) (5)

(b) (6)

By direction

Copy to: NAVAVSAFCEN (2)(Airmail) CINCPACFLT COMFAIRALAMEDA COMCARAIRGRU 21 CO, VA-212 BAR, Columbus

FF7-2/A25 Ser: 322/ **5995** 6 DEC 1957

SPECIAL HANDLING REQUIRED IN ACCORDANCE WITH PART VII OPNAV INSTRUCTION 3750.6B

THIRD ENDORSEMENT on VA-212 AAR ser 1-57 concerning FJ-4B, BuNo 141483 accident occurring 22 October 1957, pilot (b) (6)

From: Commander Fleet Air Alameda

To: Chief of Naval Operations(OP-57)

Via: (1) Commander Naval Air Force, Pacific Fleet
(2) Chief, Bureau of Aeronautics (AER-512)

(3) Director, U. S. Naval Aviation Safety Center

Subj: VA-212 AAR ser 1-57; forwarding of

1. Forwarded, concurring in the conclusions and recommendations of the Aircraft Accident Board, and in the remarks of the second endorsement, subject to the following comments:

Copy to: BUAMR (AMR-512) NAVAVSAFCEN (2) CINCPACFLT CAG-21 CO, VA-212

03

FF12/CVG-21 A25 Ser, 236 NOV 12 1957 SPECIAL HANDLING REQUIRED IN ACCORDANCE WITH PART VII OPNAV INSTRUCTION 3750.6B SECOND ENDORSEZENT on VA-212 AAR serial 1-57 concerning FJ-4B, BuNo 141483 occurring 22 October 1957, Pilot (b) (6) From: Commander Carrier Air Group TWENTY ONE Chief of Naval Operations (OP-57) (1) Commander Fleet Air Alameda Via: (2) Commander Naval Air Force, Pacific Fleet
(3) Chief, Bureau of Aeronautics (AER-512)
(4) Director, U. S. Naval Aviation Safety Center Subj: VA-212 AAR ser 1-57; forwarding of 1. Forwarded. Copy to: CINCPACELT BAR COLLEGES NAVAVSAFCEN BUARR (AER-512) COMMAVAIRPAC

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(1) Commen	der Naval Air	Force, Pac	ific Fle	es (s)En		ng Offic		Statemer	nt		
(4) Chief,	Bureau of Aer	onautics (AER-512)	(e)Sta	tement	of Towe	r Ope	erator			
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PART V THE ACCIDENT

At approximately 1555 on 22 October 1957, LTJU (b) (6) departed NAAS FALLON in FJ-4B BUNO 141483 (side number NP-211) for a schoduled loft bombing mission to target B-19. He was accompanied by a wingman, LTJG (b) (6) in FJ-4B (side number NP-209).

(b) (5), (b) (6)

SPECIAL HANDLING REQUIRED IN ACCORDANCE WITH PART VIL

exploded on impact and there were no injuries to other personnel. LTJG (b) (6) was picked up by pilots from VA-153 in a private car and later transferred to the NAAS FALLON crash ambulance. The FALLON helicopter arrived over the scene approximately 25 minutes after the pilot ejected.

PART VI DAMAGE TO ATRORAFT

As evidenced by marks on the ground and as observed by LANG (D) (6) (see enclosure 4) the aircraft made initial contact with the surface in a mear vertical bank and in an approximate 30° dive. Configuration of the aircraft was port and nose gear down, all gear doors down, starboard gear up, MK 76 bomb disper " (bread board) with 6 MK 76 practice bombs on wing station number five (mid-starboard) and a UK 7 pylon and Aero 7A bomb rack on wing station number two (mid-Port). Air speed at impact was estimated at between 270 and 300 kts. After the first contact with the surface, the aircraft traveled 30 ft on a magnetic heading of 2600 shedding parts of the port wing outer panel, at which point the aircraft exploded. The force of the explosion totally destroyed the aircraft. A crater approximately 20 feet deep and 40 feet in diameter was formed. Small pieces of wreckage and components were found 30° to each side of the heading of the aircraft beyond the point of impact. Part of the starboard outer wing panel was found 600 yards beyond the crater, which marked the limit of the scattered wreckare. The engine, tail pipe and aircraft tail section apparently buried thenselves below the crater and were not recovered. A fire burned in the crater for 36 hours following the crash:

PART VII THE INVESTIGATION

During the course of the investigation the following facts were determined:

- (1) LTJJ (b) (6) has a total of 1090.4 hours pilot time, 135.3 hours of which are in jet fighters, and 57.5 hours in FJ-4B aircraft. He has had no previous accidents or flight violations.

 (b) (5)
- (2) LTJG (b) was on an authorized flight and was cleared on a local VFR flight plan.
- (3) On the initial take off of this flight the energency one-shot nose gear extension handle had been left in the un-cocked position. This would prevent the nose gear from retracting. This handle was re-set to the cocked position prior to the second takeoff and there were no known mechanical discrepancies at this time.
- (4) Both main landing gear tires were recovered from the wreckage. One of these tires showed positive evidence that it had blown out around the inboard bend of the tire prior to the crash (see enclosure 7 print B).

This tire was identified as the port main tire upon consideration of the following points.

(a) All other tires that suffered blow-outs in the week immediately preceding this accident and those subsequent, blew-out on the inboard side around the bead. There were a total of seven similar blow-outs since 11 Oct. 1957 including one subsequent to this accident.

(b) This tire was called with and and was located only 50 feet from the crater. The other tire was clean and located approximately 200 yards from the crater. Since the port year was down, it follows that this gear contacted the ground prior to explosion and thereby picked up the mud while the other tire was retracted and probably was carried to its final resting place by the explosion.
(c) Following point (a) above, the feather-edging of the tread of the tire in question matched the markings of port tires installed on other squadron aircraft. This feather-edging of the tread is characteristic on all tires after use and occurs because of the tendency of the rubber to be pulled back during

braking. The resultant merking of the treed is always a alight roughness

which points in a direction op osite to rotation.

(b)(5)

- (6) In the ten days preceding this accident, this squadron suffered five of er cases of blown tires after the aircraft had been landed. In two cases the tires exploded on the line 5 to 10 minutes after landing and in the other cases, the tires exploded during taxi-in or taxi-out to takeoff. On the day following the accident all tires were removed from all squadron aircraft and inspected. Eleven out of 18 tires were rejected due to obvious signs of deterioration due to over-heating. In every case the tires were of the same assurfactor and lot number.
- (7) The type of operations being conducted at the time have a bearing on the accident, particularly the following: Field elevation 4000 feet, taxi distance back to the line after landing of approximately 2 miles, runmay temperatures of 70-00 F., take off without shutting down, with about 15 to 20 minutes on the deck between landing and takeoff. External loading of the aircraft during these operations were 6 kK 76 practice bombs and a kK 76 bomb dispenser. There was no asymmetrical loading.

(b) (5)

(11) According to maintenance records this aircraft completed a second intermediate inspection on the morning of the day the accident occurred. During this inspection the engine was changed and the brakes, wheels and tires were replaced. Following the inspection the aircraft was flown on a test flight of 0.9 hours duration. A high RPM and low ROT discrepancy was reported after this flight which was corrected prior to the next flight.

PART VIII THE ANALYSIS

(b) (5)

IN ACCORDANCE WITH PART VII OPNAV INST. 3750.68

	(b) (5)				
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PART IN CONCLUSIONS AN RECOMMENDATIONS

(b) (5)

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(b) (5), (b) (6)

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SPECIAL HANDLING REQUIRED IN ACCORDANCE WITH PART VIII OPNAY INST. 3750.68

AIRCRAFT ACCIDENT REPORT NAAS FAIION, NEVADA

BUREAU NO 141	183_TYPE	FJ-4B	PILOT	'S NAME	(b) (b)	LT.
UNIT ATTACHED	VA-212	_TDME_	1639 PST		DATE	22 OCT. 1952
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